

## Handout: Channel Flow - Definitions and Governing Equations

### Definition

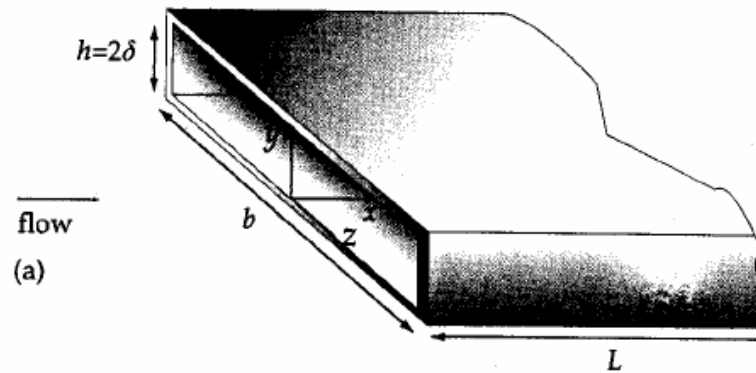


Fig. 1 Sketch of channel flow (Pope's book)

### Assumption:

$$L \gg \delta$$

$$b \gg \delta$$

We focus on the fully developed region (large  $x$ ) in which velocity statistics no longer vary with  $x$ . As the duct has also a large aspect ratio, the turbulence statistics are homogeneous in  $x$  and  $z$  (away from the lateral walls)

The lower and upper walls are situated at  $y=0$  and  $y=2\delta$  with the channel centerline situated at  $y=\delta$

### Reynolds numbers:

$$Re = \frac{2\delta\bar{U}}{\nu}; \quad \bar{U} \text{ is bulk velocity}$$

$$Re_0 = \frac{\delta U_0}{\nu}; \quad U_0 = \langle U \rangle_{y=\delta} \text{ is the centerline velocity,}$$

$\langle \rangle$  denotes Reynolds averaged quantity

Laminar flow for  $Re < 1350$ , Fully developed for  $Re > 1800$ , transitional effects present up to  $Re = 3000$

## Governing Equations

### Continuity

$$\frac{\partial \langle U \rangle}{\partial x} + \frac{\partial \langle V \rangle}{\partial y} = 0$$
$$\Rightarrow \frac{\partial \langle V \rangle}{\partial y} = 0$$

Integrate in y with  $\langle V \rangle_{y=0} = 0$

$$\Rightarrow \langle V \rangle = 0$$

### Lateral Momentum Equation

$$\langle U \rangle \frac{\partial \langle V \rangle}{\partial x} + \langle V \rangle \frac{\partial \langle V \rangle}{\partial y} = -\frac{1}{\rho} \frac{\partial \langle p \rangle}{\partial y} + \nu \frac{\partial^2 \langle V \rangle}{\partial x^2} + \nu \frac{\partial^2 \langle V \rangle}{\partial y^2} - \frac{\partial \langle uv \rangle}{\partial x} - \frac{\partial \langle v^2 \rangle}{\partial y}$$
$$\Rightarrow \frac{\partial}{\partial y} \left( \frac{\langle p \rangle}{\rho} + \langle v^2 \rangle \right) = 0$$

Integrate from 0 to y

$$\langle v^2 \rangle + \frac{\langle p \rangle}{\rho} = \frac{p_w(x)}{\rho}$$

Differentiate with respect to x

$$\frac{\partial \langle p \rangle}{\partial x} = \frac{\partial p_w}{\partial x}$$

$\Rightarrow$  Pressure gradient uniform across flow

### Axial Momentum Equation

$$\langle U \rangle \frac{\partial \langle U \rangle}{\partial x} + \langle V \rangle \frac{\partial \langle U \rangle}{\partial y} = -\frac{1}{\rho} \frac{\partial \langle p \rangle}{\partial x} + \nu \frac{\partial^2 \langle U \rangle}{\partial x^2} + \nu \frac{\partial^2 \langle U \rangle}{\partial y^2} - \frac{\partial \langle u^2 \rangle}{\partial x} - \frac{\partial \langle uv \rangle}{\partial y}$$

Total shear stress (viscous + Reynolds stress):

$$\tau = \rho \nu \frac{\partial \langle U \rangle}{\partial y} - \rho \langle uv \rangle$$

$$\Rightarrow \frac{\partial \tau}{\partial y} = \frac{\partial p_w}{\partial x}$$

$\Rightarrow$  Total shear stress compensated by pressure gradient

$$\frac{\partial \tau}{\partial y} \neq f(x) \text{ and } \frac{\partial p_w}{\partial y} \neq f(y)$$

$$\Rightarrow \frac{\partial \tau}{\partial y} = \frac{\partial p_w}{\partial x} = \text{const}$$

$$\tau(y) = \tau_w \left( 1 - \frac{y}{\delta} \right)$$

Skin friction coefficients

$$c_f \equiv \frac{\tau_w}{1/2 \rho U_0^2}$$

$$C_f \equiv \frac{\tau_w}{1/2 \rho \bar{U}^2}$$

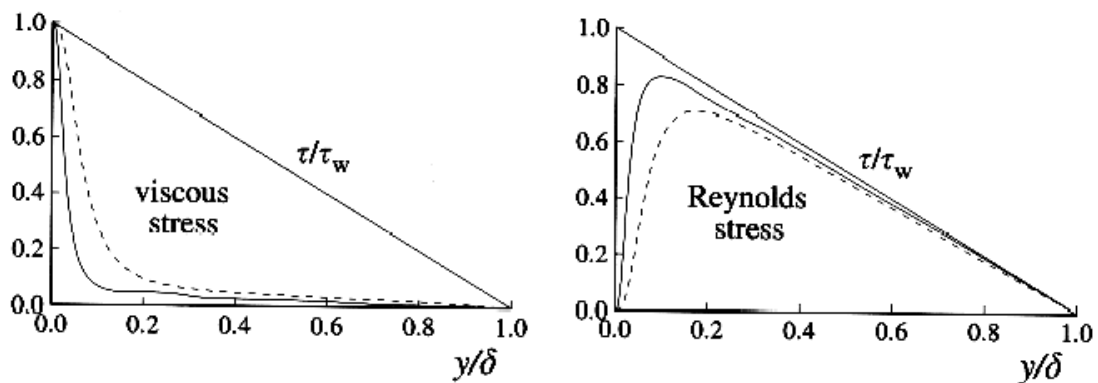


Fig. 7.3. Profiles of the viscous shear stress, and the Reynolds shear stress in turbulent channel flow: DNS data of Kim *et al.* (1987): dashed line,  $Re = 5,600$ ; solid line,  $Re = 13,750$ .

Figure 7.3 from Pope's Turbulent Flow book: Reynolds stress, viscous stress, and total stress (DNS of KIM, Mion, Moser (1987))

## Wall Shear Stress

$\tau_w$  given only by viscous stress as  $\langle uv \rangle = 0$  at the wall

$$\tau_w = \rho \nu \left. \frac{\partial \langle U \rangle}{\partial y} \right|_{y=0}$$

Friction velocity:

$$u_\tau \equiv \sqrt{\frac{\tau_w}{\rho}}$$

Viscous length scale:

$$\delta_\nu \equiv \frac{\nu}{u_\tau} = \nu \sqrt{\frac{\rho}{\tau_w}}$$

$$\Rightarrow \text{Re}_\nu \equiv \frac{\delta_\nu u_\tau}{\nu} = 1$$

Friction Reynolds number (Reynolds number defined with the friction velocity):

$$\text{Re}_\tau = \frac{u_\tau \delta}{\nu} = \frac{\delta}{\delta_\nu}$$

Wall units:

$$y^+ = \frac{y}{\delta_\nu} = \frac{u_\tau y}{\nu} \quad (\text{similar to local Reynolds number})$$

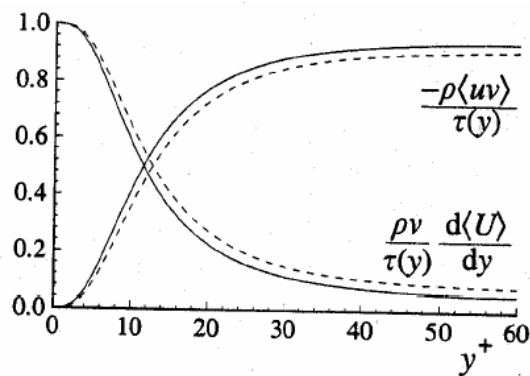


Fig. 7.4. Profiles of the fractional contributions of the viscous and Reynolds stresses to the total stress. DNS data of Kim *et al.* (1987): dashed lines,  $\text{Re} = 5,600$ ; solid lines,  $\text{Re} = 13,750$ .

- Outer layer:  $y^+ > 50$  (viscous stresses can be neglected, direct effect of viscosity is negligible)

- Viscous wall region:  $y^+ < 50$
- Viscous sublayer:  $y^+ < 5$  (Reynolds stresses are negligible)

As  $Re$  increases, the fraction of the channel occupied by the viscous wall region decreases, since  $\frac{\delta_v}{\delta} \approx Re_\tau^{-1}$